

# Structures Key Equations

## Static Determinacy

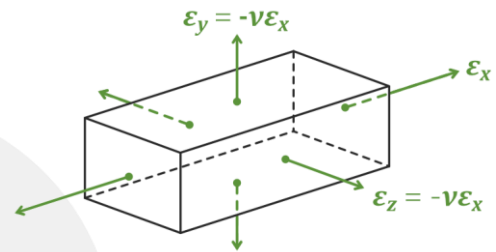
Mechanism	$b + n < 2j$
May be determinate	$b + n = 2j$
Indeterminate	$b + n > 2j$

- $b$  is the number of bars
- $n$  is the number of reactions
- $j$  the number of pins

## Stress & Strain

Engineering Stress	$\sigma = \frac{F}{A}$
Engineering Strain	$\varepsilon = \frac{x}{L}$
Young's Modulus	$E = \frac{\sigma}{\varepsilon}$
Poisson's Ratio	$\varepsilon_y = \varepsilon_z = -\nu\varepsilon_x$
Hydrostatic Stress	$\sigma_H = \sigma_x = \sigma_y = \sigma_z$

## Rectangular Cross-Sections



Change in Volume	$\Delta V = \varepsilon_V \times V$
Volumetric Strain	$\varepsilon_V = \frac{\Delta V}{V} = \varepsilon_x + \varepsilon_y + \varepsilon_z$

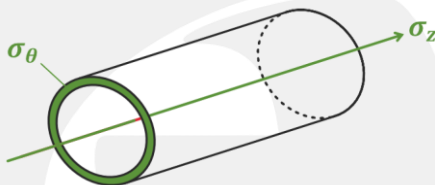
Bulk Modulus	$K = \frac{\sigma_H}{\varepsilon_V}$
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$$\varepsilon_x = \frac{1}{E} (\sigma_x - \nu(\sigma_y + \sigma_z)) + \alpha\Delta T$$

$$\varepsilon_y = \frac{1}{E} (\sigma_y - \nu(\sigma_x + \sigma_z)) + \alpha\Delta T$$

$$\varepsilon_z = \frac{1}{E} (\sigma_z - \nu(\sigma_x + \sigma_y)) + \alpha\Delta T$$

## Cylindrical Thin-Walled Pressure Vessels



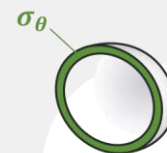
Hoop Stress, $\sigma_\theta$	$\sigma_\theta = \frac{PR_i}{t}$
Axial Stress, $\sigma_z$	$\sigma_z = \frac{\sigma_\theta}{2} = \frac{PR_i}{2t}$
Axial strain, $\varepsilon_z$	$\varepsilon_z = \frac{\Delta L}{L}$
Hoop strain, $\varepsilon_\theta$	$\varepsilon_\theta = \frac{\Delta r}{r}$
Volumetric Strain	$\varepsilon_V = \frac{\Delta V}{V} = 2\varepsilon_\theta + \varepsilon_z$

$$\varepsilon_z = \frac{1}{E} (\sigma_z - \nu(\sigma_\theta + \sigma_r)) + \alpha\Delta T$$

$$\varepsilon_\theta = \frac{1}{E} (\sigma_\theta - \nu(\sigma_r + \sigma_z)) + \alpha\Delta T$$

$$\varepsilon_r = \frac{1}{E} (\sigma_r - \nu(\sigma_\theta + \sigma_z)) + \alpha\Delta T$$

## Spherical Thin-Walled Pressure Vessels



Hoop Stresses, $\sigma_\theta, \sigma_\phi$	$\sigma_\theta = \sigma_\phi = \frac{PR_i}{2t}$
Hoop Strains, $\varepsilon_\theta, \varepsilon_\phi$	$\varepsilon_\theta = \varepsilon_\phi = \frac{\Delta r}{r}$
Volumetric Strain	$\varepsilon_V = \frac{\Delta V}{V} = 3\varepsilon_\theta$

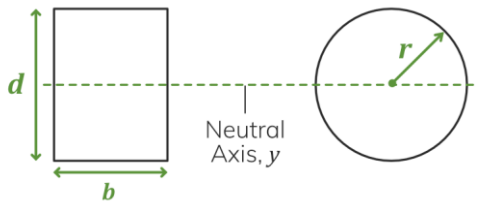
$$\varepsilon_\theta = \frac{1}{E} (\sigma_\theta - \nu(\sigma_r + \sigma_\phi)) + \alpha\Delta T$$

$$\varepsilon_\phi = \frac{1}{E} (\sigma_\phi - \nu(\sigma_r + \sigma_\theta)) + \alpha\Delta T$$

$$\varepsilon_r = \frac{1}{E} (\sigma_r - \nu(\sigma_\theta + \sigma_\phi)) + \alpha\Delta T$$

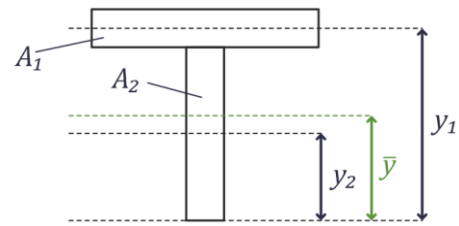
- Take radial stresses and strains as zero in both spheres and cylinders.
- Axial strain in a sphere is also zero.

## Second Moment of Area



$$I = \frac{bd^3}{12}$$

$$I = \frac{\pi r^4}{4}$$



$$\bar{y}(A_{total}) = y_1(A_1) + y_2(A_2)$$

$$I_{xx} = [I_1 + A_1(y_1 - \bar{y})^2] + [I_2 + A_2(\bar{y} - y_2)^2]$$

## Beam Theory

Key Relation

$$\frac{\sigma}{y} = \frac{M}{I} = \frac{E}{R}$$

Slope Angle

$$\frac{dv}{dx} = \frac{1}{EI} \int M dx$$

Radius & Moment

$$R = \frac{d^2v}{dx^2} = \frac{1}{EI} M$$

Deflection

$$v = \frac{1}{EI} \iint M dx$$

## Torsion

Key Relation

$$\frac{\tau}{r} = \frac{T}{J} = \frac{G\theta}{L}$$

Torsional Stiffness,  $K_T$

$$K_T = \frac{T}{\theta} = \frac{JG}{L}$$

Shear Stress & Strain

$$\gamma = \frac{\tau}{G}$$

$$\frac{1}{K_{T,total}} = \frac{1}{K_{T1}} + \frac{1}{K_{T2}} \dots$$

Shear Modulus

$$G = \frac{E}{2(1 + \nu)}$$

## Torsion in Thin-Walled Shafts

Shear Strain

$$\gamma = \frac{GR_0}{L}$$

Shear Stress

$$\tau = \frac{GR_0\theta}{L}$$

$$\tau = \frac{T}{2\pi R_0^2 t}$$

## Torsion in Solid & Hollow Shafts

Shear Strain

$$\gamma(r) = \frac{r\theta}{L}$$

Shear Stress

$$\tau = \frac{Gr\theta}{L}$$

Torque

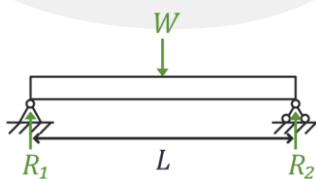
$$T = \frac{G\theta J}{L}$$

2<sup>nd</sup> Polar Moment of Area

$$J = \frac{\pi D^4}{32}$$

## Standard Slope & Deflection of Beams

Simply Supported with Point Mass



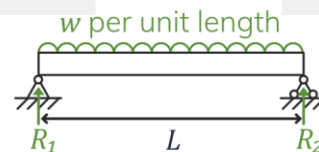
End Slope

$$\frac{WL^2}{16EI}$$

Central Deflection

$$\frac{WL^3}{48EI}$$

Simply Supported with Distributed Load



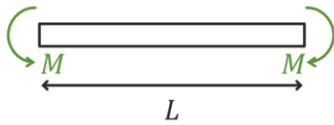
End Slope

$$\frac{wL^3}{24EI}$$

Central Deflection

$$\frac{5wL^4}{384EI}$$

### Moment Supported



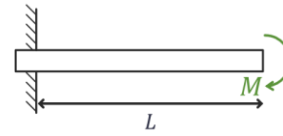
End Slope

$$\frac{ML}{2EI}$$

Central Deflection

$$\frac{ML^2}{8EI}$$

### Cantilever with Moment



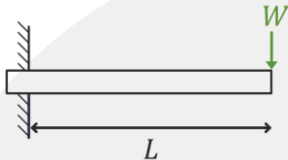
End Slope

$$\frac{ML}{EI}$$

End Deflection

$$\frac{ML^2}{2EI}$$

### Cantilever with Point Load



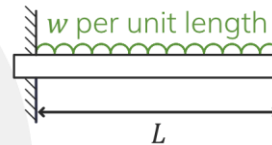
End Slope

$$\frac{WL^2}{2EI}$$

End Deflection

$$\frac{WL^3}{3EI}$$

### Cantilever with Distributed Load



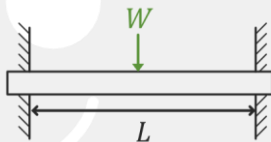
End Slope

$$\frac{wL^3}{6EI}$$

End Deflection

$$\frac{wL^4}{8EI}$$

### Built in at Both Ends with Central Point Load



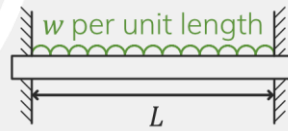
End Moment

$$\frac{WL}{8}$$

Central Deflection

$$\frac{WL^3}{192EI}$$

### Built in at Both Ends with Distributed Load



End Moment

$$\frac{wL^2}{12}$$

Central Deflection

$$\frac{wL^4}{384EI}$$

### Principal Stresses

Direct stresses

$$S_N = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta$$

Principal stresses

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} + \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2}$$

Shear stresses

$$S_S = \frac{\sigma_x - \sigma_y}{2} \sin 2\theta - \tau_{xy} \cos 2\theta$$

Direct direction

$$\tan 2\theta_P = \frac{2\tau_{xy}}{\sigma_x - \sigma_y}$$

Shear direction

$$\tan 2\theta_S = \frac{-(\sigma_x - \sigma_y)}{2\tau_{xy}}$$

- $\theta_P$  and  $\theta_S$  are always  $45^\circ$  apart

### Maximum Shear Stresses, $\hat{t}$

One of  $\sigma_1, \sigma_2$  positive, one negative  $\hat{t} = \frac{\sigma_1 - \sigma_2}{2}$

Both  $\sigma_1, \sigma_2$  positive

$$\hat{t} = \frac{\sigma_1}{2}$$

Both  $\sigma_1, \sigma_2$  negative

$$\hat{t} = \frac{\sigma_2}{2}$$

Tresca Failure Criterion

$$\hat{t} = \max \left( \left| \frac{\sigma_1 - \sigma_2}{2} \right|, \left| \frac{\sigma_2 - \sigma_3}{2} \right|, \left| \frac{\sigma_1 - \sigma_3}{2} \right| \right)$$

Von Mises (Strain Energy) Failure Criterion

$$\hat{t} = \sqrt{\frac{1}{3} \left[ \left( \frac{\sigma_1 - \sigma_2}{2} \right)^2 + \left( \frac{\sigma_2 - \sigma_3}{2} \right)^2 + \left( \frac{\sigma_3 - \sigma_1}{2} \right)^2 \right]}$$

### Principal Strains

Direct strains

$$e_n = \frac{e_x + e_y}{2} + \frac{e_x - e_y}{2} \cos 2\theta + \frac{\gamma_{xy}}{2} \sin 2\theta$$

Principal Strains

$$e_{1,2} = \frac{e_x + e_y}{2} \pm \frac{1}{2} \sqrt{(e_x - e_y)^2 + \gamma_{xy}^2}$$

### Maximum Shear Strains

$$e_{s \max, \min} = \pm \frac{e_1 - e_2}{2}$$

Shear strains

$$\frac{e_s}{2} = \frac{e_x - e_y}{2} \sin 2\theta - \frac{\gamma_{xy}}{2} \cos 2\theta$$

Direct strain direction

Shear direction

$$\tan 2\theta_p = \frac{\gamma_{xy}}{e_x - e_y}$$

Type equation here.

### Finding $\sigma_x$ from shear stresses and strains

$$\sigma_x = \frac{E(e_x + \nu e_y)}{1 - \nu^2}$$